

Serial No. Not Yet Assigned
Atty. Doc. No. 2003P07721WOUS

Amendments To The Specification:

In the English translation document, please delete the term --Description-- at page 1 line 1, before the title.

In the English translation document, please add the paragraph at page 1 line 5, after the title, as follows:

--CROSS REFERENCE TO RELATED APPLICATIONS

This application is the US National Stage of International Application No. PCT/EP2004/007945, filed July 16, 2004 and claims the benefit thereof. The International Application claims the benefits of European Patent application No. 0301837.8 EP filed August 13, 2003, all of the applications are incorporated by reference herein in their entirety.--

In the English translation document, please amend the paragraph beginning at page 1 line 6, after the newly added CROSS REFERENCE TO RELATED APPLICATIONS section, as follows:

--FIELD OF THE INVENTION

The invention relates to a method for braking a rotor of a turbine engine according to the ~~preamble of claim 1~~ claims and to a turning gear for driving a rotor of a turbine engine according to the ~~preamble of claim 6~~ claims. --

In the English translation document, please insert the section heading at page 1 line 10, as follows:

--BACKGROUND OF THE INVENTION--

In the English translation document, please insert the section heading at page 2 line 23, as follows:

--SUMMARY OF THE INVENTION--

In the English translation document, please amend the paragraph beginning at page 5 line 1, as follows:

--BRIEF DESCRIPTION OF THE DRAWINGS

The invention is explained with reference to ~~a drawing~~ drawings in which:

Figure 1 shows a ~~diagrammatic illustration of a turbine engine with a turning gear longitudinal part section through a gas turbine,~~

Figure 2 shows a ~~longitudinal part section through a gas turbine~~ diagrammatic illustration of a turbine engine with a turning gear--

In the English translation document, please amend the paragraph beginning at page 5 line 9, as follows:

--DETAILED DESCRIPTION OF THE INVENTION

Figure 2 shows a gas turbine 1 in a longitudinal part section. It has, inside, a rotor 3 which is rotary-mounted about an axis of rotation 2 and which is also designated as a turbine rotor or rotor shaft. An intake casing 4, a compressor 5, a toroidal annular combustion chamber 6 having a plurality of coaxially arranged burners 7, a turbine 8 and the exhaust gas casing 9 succeed one another along the rotor 3.--

In the English translation document, please amend the paragraph beginning at page 6 line 7, as follows:

--When the gas turbine 1 is in operation, air 21 is sucked in through the intake casing 4 by the compressor 5 and is compressed in the compressor duct 10. The air 21 provided at that end of the compressor 5 which is on the burner side is led through the diffuser 11 to the burner 7 and is mixed there with a fuel. The mixture is then burnt in the combustion space 10 so as to form a working fluid 20. The working fluid 20 flows from there into the hot-gas duct 13. The working fluid 20 expands at the guide blades 16 arranged in the turbine 8 and at the moving blades 18 so as to transmit pulses, so that the rotor 3 is driven and, together with the latter, a working machine (not illustrated) coupled to it.--